National Transportation Safety Board Washington, DC 20594

Brief of Accident

Adopted 04/01/2003

DEN02	LA101
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File No. 12761	08/30/2002	Canon City, CO	Aircraft Reg No. N43GA		Time (Local): 12:25 MDT	
Make/Model: Engine Make/Model: Aircraft Damage: Number of Engines: Operating Certificate(s): Type of Flight Operation: Reg. Flight Conducted Under:	Destroyed 1 None Personal		Crew Pass	Fatal 0 0	Serious 1 0	Minor/None 0 0
Destination:	Same as Accident/Incident Locati Local Flight Off Airport/Airstrip	on	Condition of Light: Day Weather Info Src: Weather Observation Facility Basic Weather: Visual Conditions Lowest Ceiling: None Visibility: 10.00 SM Wind Dir/Speed: / 006 Kts Temperature (°C): 29 Precip/Obscuration: None / None			
Pilot-in-Command Age:	57			Flight Ti	me (Hours)	
Certificate(s)/Rating(s) Airline Transport; Single-engine Land; Helicopter Instrument Ratings None			Total All Aircraft: 20000 Last 90 Days: Unk/Nr Total Make/Model: 5000 Total Instrument Time: 250			

The pilot said that he had been flying down in the canyon, and had come up to the surface to return to his helipad. He said that the helicopter's mast struck a wire, and severed the main rotor pitch change tubes disabling the aircraft. The pilot said that the helicopter came down vertically (20 to 30 feet) breaking the left skid, crushing the fuselage, separating the tail boom, and separating portions of both main rotor blades.

Brief of Accident (Continued)

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Occurrence #1: IN FLIGHT COLLISION WITH OBJECT

Phase of Operation: MANEUVERING

Findings

1. OBJECT - WIRE, STATIC

2. (C) CLEARANCE - NOT MAINTAINED - PILOT IN COMMAND

3. (C) ROTORCRAFT FLIGHT CONTROL - CUT/SEVERED

Occurrence #2: IN FLIGHT COLLISION WITH TERRAIN/WATER

Phase of Operation: DESCENT - UNCONTROLLED

Findings

4. (C) AIRCRAFT CONTROL - NOT POSSIBLE - PILOT IN COMMAND

5. TERRAIN CONDITION - MOUNTAINOUS/HILLY

Findings Legend: (C) = Cause, (F) = Factor

The National Transportation Safety Board determines the probable cause(s) of this accident as follows. the pilot's failure to maintain clearance of a static wire, and the subsequent loss of aircraft control due to the main rotor pitch change tubes being severed.